

A vertical strip on the left side of the slide shows a satellite view of Earth from space. It captures the curvature of the planet, with a thin blue line of the atmosphere separating the dark, starry void of space from the bright, cloud-covered surface of the Earth. Below the clouds, the glowing yellow and orange lights of cities and urban areas are visible, particularly concentrated along a coastline.

Airbus Bartolomeo Platform

General Overview



Airbus Bartolomeo Platform

- External Platform on the International Space Station (ISS)
- Attached to Columbus Module
- Wide Range of Mission Opportunities
- Payload alignment in various directions (Ram, Zenith, Nadir)
- Provides highest data downlink rate on ISS
- Expected launch on SpX-20 (March 2020)
- Installation completed in April 2020



Airbus Bartolomeo: Robust Capabilities

- Capability to take advantage of all 3 major LEO phenomena:
 - Extreme Conditions of Space
 - Vantage Point - Remote Sensing and Satellite Deployment
 - Microgravity

- Ability to service ISS NL major markets and OGA Programs
 - Technology Development and TRL
 - Advanced Materials
 - Remote Sensing
 - NSF Transport Solicitation

Use Cases- Advanced Materials & Manufacturing

Use Case	Platform Capabilities & Benefits
Materials Testing	<ul style="list-style-type: none">▪ With unobstructed Zenith-oriented view Bartolomeo gives the opportunity to expose material samples to space and solar radiation▪ With unobstructed Ram-facing view the effects of atomic oxygen can be studied on samples▪ Additional exposure to elements such as thermal cycling, vacuum, and space debris are present in all directions
In-Space Manufacturing	<ul style="list-style-type: none">▪ Via Bartolomeo and its large / extendable payload envelopes on orbit in-space manufacturing can be performed to produce structures with fewer defects via 3D printing or other appropriate methods

Use Cases- Remote Sensing & Monitoring

Use Case	Platform Capabilities & Benefits
Remote Sensing	<ul style="list-style-type: none">▪ The unobscured view of Earth from Bartolomeo in approximately 400 km orbit altitude enables high quality imaging with cost-efficient instrumentation▪ Types of sensing can include (IR, Visible, UV, RF, etc.)
Astrophysics & Heliophysics	<ul style="list-style-type: none">▪ Bartolomeo offers among the best view towards the Zenith direction from the ISS
Atmospheric Research	<ul style="list-style-type: none">▪ All forward-facing payloads have unobstructed view to the space / atmosphere boundary▪ Broadband data downlink capabilities of Bartolomeo allows for a high data production rate
Space Weather	<ul style="list-style-type: none">▪ The unobstructed Zenith-oriented view allows cost-efficient space observation, e. g. for solar activity monitoring

Use Cases- Technology Development & Testing

Use Case	Platform Capabilities & Benefits
On-Orbit Assembly for Exploration	<ul style="list-style-type: none">▪ Bartolomeo can provide an opportunity to assemble space system components on-orbit and deploy them with appropriate systems▪ Short-term realization of a long-term vision to provide larger space systems unrestricted by the launcher payload fairing for exploration
Robotics Testing	<ul style="list-style-type: none">▪ Bartolomeo can provide an opportunity to perform robotic operations in a protected testing environment
In-orbit Testing	<ul style="list-style-type: none">▪ With power, data and viewing available Bartolomeo can serve as general in-orbit demonstration test bed▪ If compliant with safety regulations any technology can be tested on ISS as long as it is of civilian purpose
Propulsion Testing	<ul style="list-style-type: none">▪ With available power of up to 800 W per payload Bartolomeo can serve as testbed for new electric space propulsion systems
Spacecraft Deployment	<ul style="list-style-type: none">▪ One of the Bartolomeo payload sites can be converted to a small satellite deployment system▪ If deployed directly from Bartolomeo satellites can have more mass than deployable by existing systems on the ISS

NSF/CASIS Transport Phenomena Solicitation Focus Areas for Bartolomeo		
Main Solicitation Focus Area	Specific Focus Area	Types of Investigations for Bartolomeo Platform
Thermal Transport	Radiation	<ul style="list-style-type: none"> ▪ Payloads on the Bartolomeo platform can be exposed to the external environment of space which contains high levels of radiation ▪ Platform slots can face three different directions (ram, zenith, nadir) where each provide a different level of radiation exposure on a payload
Thermal Transport & Fluid Dynamics	Phase Transitions	<ul style="list-style-type: none"> ▪ Studies that can utilize both the microgravity and the rapid temperature cycling of the external space environment
Thermal Transport & Fluid Dynamics	High Resolution Modeling	<ul style="list-style-type: none"> ▪ Bartolomeo downlink can deliver larger amounts of data faster than current ISS capabilities, improving the real time modeling capabilities for thermal profiles, single & multiphase flow profiles, etc.
Fluid Dynamics	Cryogenics	<ul style="list-style-type: none"> ▪ Bartolomeo can support low temperature investigations and is safer to use with more volatile fluids being an external platform
Combustion	Low Temperature Chemistry	<ul style="list-style-type: none"> ▪ Lower temperature combustion chemistry can be supported on the platform and is safer to conduct externally



Bartolomeo Services

- **Included:**

- Payload Mission Manager for guidance through agency acceptance and safety reviews
- Final payload tests, integration and installation on the launcher
- Launch (USA and ESA countries)
- Installation on Bartolomeo Platform
- Payload-Bartolomeo interface
- Operation interface via the Airbus Cloud Console
- Data Delivery
- End of Life Disposal

- **Extra Options:**

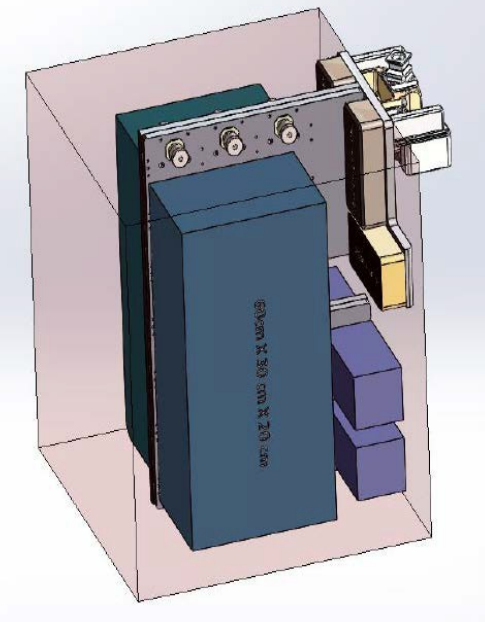
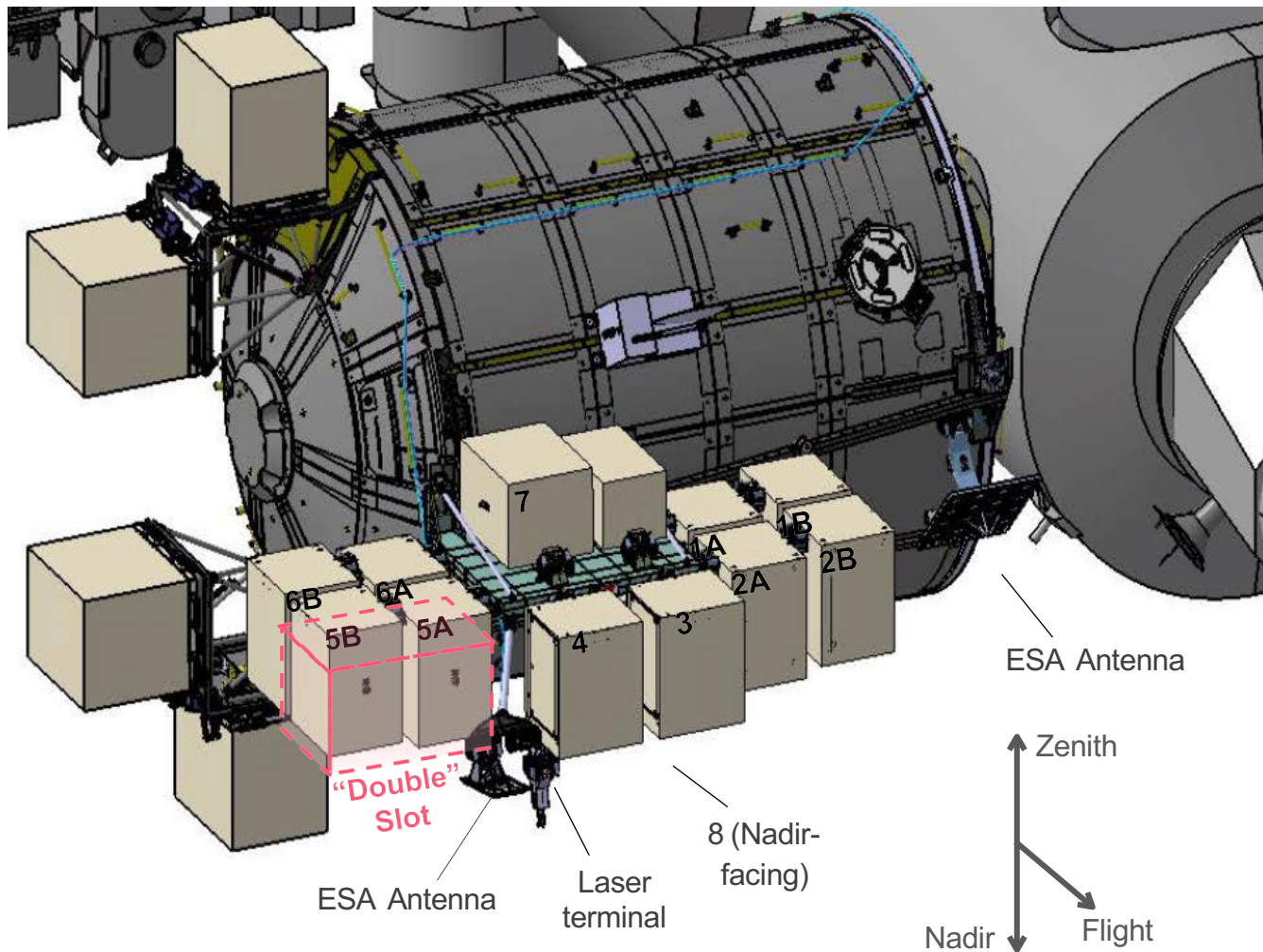
- Up to 2 TB/day data provision via Bartolomeo's own optical link (laser comm)
- Payload development consultancy
- Payload return to Earth



Bartolomeo Specifications

- **12 total slots on platform:**
 - Volume per slot: up to $100 \times 80 \times 70 \text{ cm}^3$ (560 U)
 - Slots may be subdivided for small payloads, or doubled for larger payloads
- **Mass per slot: up to 250 kg**
- **Power per slot:**
 - Up to 180 W (standard) – supports power requirements of many sensors, telescopes
 - Up to 800 W (specific slots) – for furnaces (heaters), manufacturing systems, propulsion tests
- **Data rate:**
 - Up to 0.1 Mbit/s (via ISS)
 - Up to 2 TB/day ($\sim 185 \text{ Mbit/s}$) via Bartolomeo optical link (extra)





Payload Accommodation



ArgUS Multi-Payload Adapter

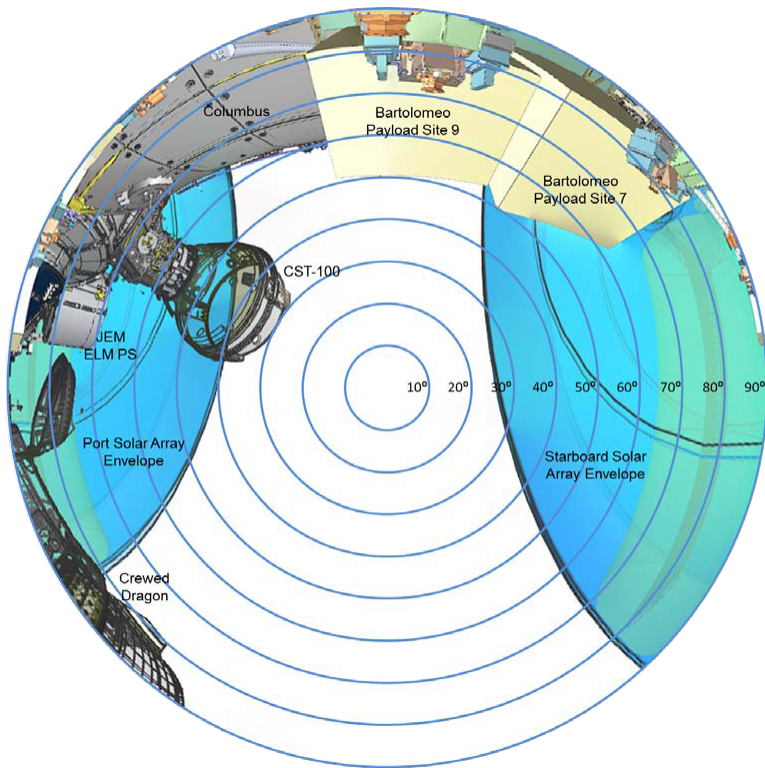
- Up to 120 U on each side
- Can hold 10 separate payloads

Payload Accommodation

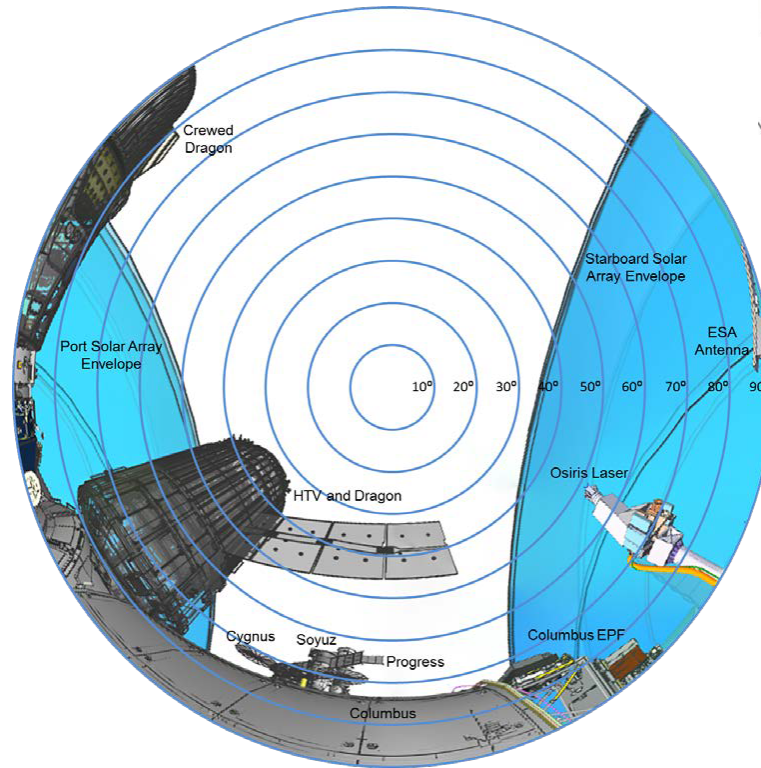
SLOT POSITION		1A	1B	2A	2B	3	4	5A	5B	6A	6B	7	8
[Option to combine slots]													
Power	800 W			x		x	x	x					
	400 W	x								x		x	x
	180 W		x		x				x		x		
Field of View	[Ram]			x	x	x	x	x	x				
	[Nadir]	x	x	x	x	x	x	x	x	x	x		x
	[Zenith]	x	x	x	x	x	x	x	x	x	x	x	
	[Port/Starboard]		x		x	x	x		x		x	x	x
[Operation of safety-critical payloads enabled]				x	x	x	x	x		x			

- > Some payload slots can be combined into "Double Slots"
- > High power 400–800 W can be provided
- > Unconstrained field of view in Ram, Nadir, Zenith
- > Some payload slots allow the operation of safety-critical payloads

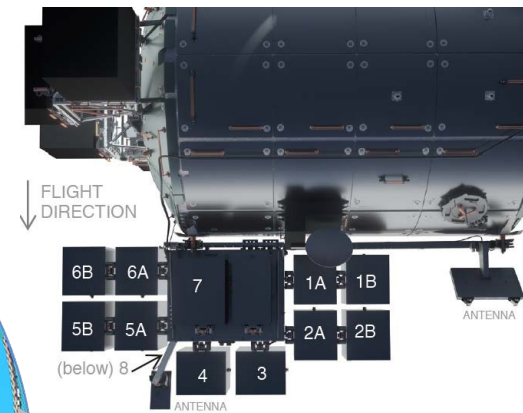
Payload Fields of View



Slot 3 Zenith View
[image credit: NASA]



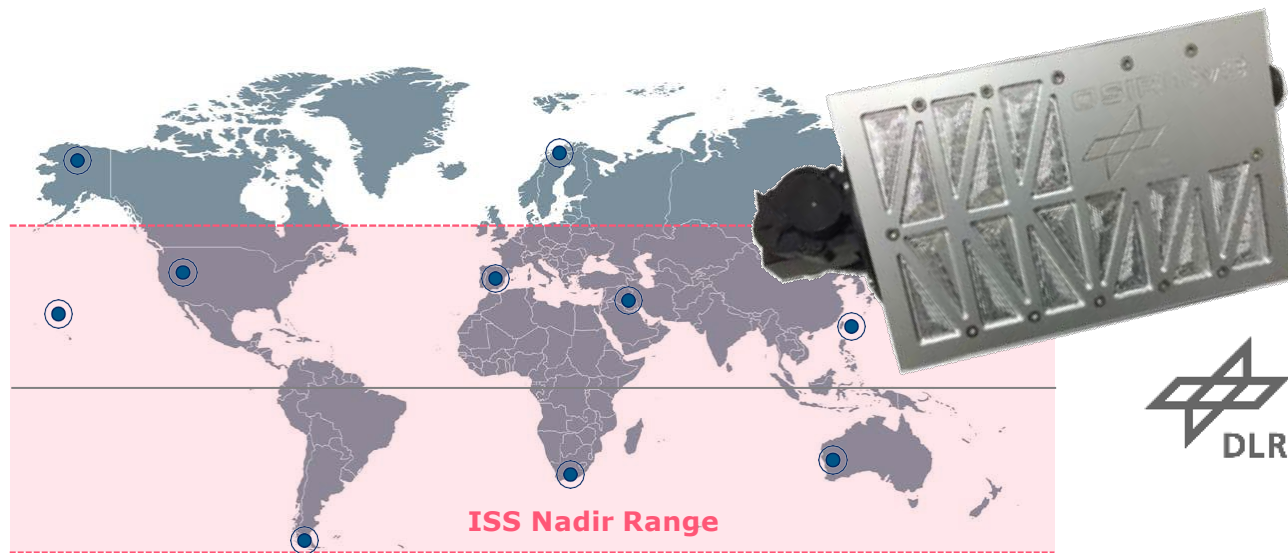
Slot 3 Nadir View
[image credit: NASA]



Payload viewing quality

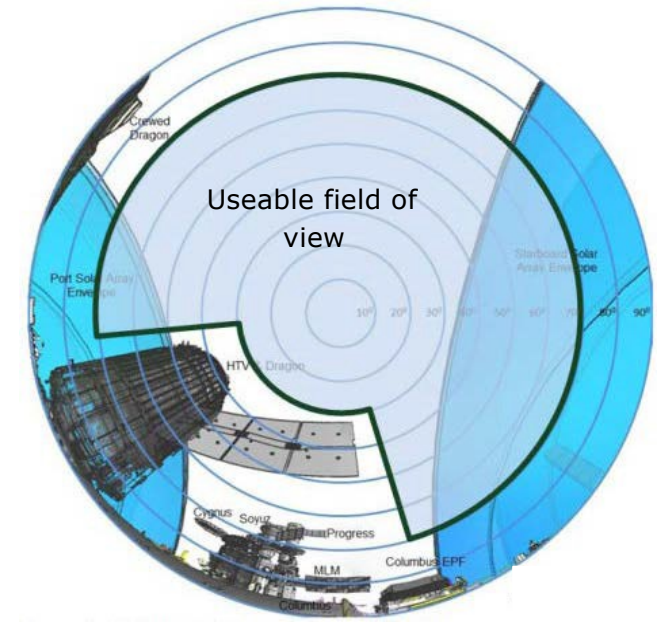
Slots	Nadir	Zenith	Ram
1A			
1B			
2A			
2B			
3			
4			
5A			
5B			
6A			
6B			
7			
8			

Broadband Data Downlink



Optical Ground Segment

- Commercial ground stations distributed worldwide will be used
- Some field of view restrictions exist to visiting vehicles and solar arrays
- Conservative achievable mean data throughput 1.375 TB / day, maximum achievable 2.5 TB / day
- Ground segment can provide around 100 GB within 45 minutes of downlink



OSIRISv3 Field of View

Throughput Analysis Parameters

OSIRISv3 max. channel rate	10 Gbps
Buffer size	500 GB
Number of ground stations	8
Minimum elevation angle	15 deg

Mapping of Uses Cases to Platform Positions

Slot Direction

Use - > Case	Materials Testing	In-Space Manufacturing	Remote Sensing	Astro & Helio Physics	Atmo-spheric Research	Space Weather	On-Orbit Assembly	Robotics Testing	In-Orbit Testing	Propulsion Testing	Spacecraft Deployment
Nadir (All except 7)		X	X				X	X	X		
Zenith (All except 8)	X	X		X		X	X	X	X	X	X
Ram (2B, 3, 4, 5B, 7)	X	X			X		X	X	X	X	X